

# Orientus, Spatial & Certus Allan Variance





## **REVISION HISTORY**

VERSION	DATE	CHANGES
1.1	5th April 2024	Added Accelerometer Allan Variance
1.0	4th April 2024	Initial Release

Table 1. Revision history

### INTRODUCTION

Orientus, Spatial and Certus are micro-electromechanical systems (MEMS) inertial measurement and/or inertial navigation units (IMU/INS). Each of these units contains the same MEMS gyroscopes and accelerometers.

This report uses the Allan Variance method to demonstrate bias instability and random walk values for both the gyroscopes and accelerometers. This is useful for comparing the raw sensor performance of an AHRS or INS product. This report facilitates decision making in the suitability and performance of Orientus, Spatial & Certus for a wide range of applications.

For further application, performance, or reference needs, please contact your account manager or <a href="mailto:support@advancednavigation.com">support@advancednavigation.com</a> for more information.



# **ALLAN VARIANCE**

The Allan Variance method is used to determine the Bias instability and Angular/Velocity Random Walk (ARW/VRW) of the gyroscopes and accelerometers.

**Bias Instability** (sometimes referred to by others as Bias Stability, or In-run Bias Stability) is a comparative figure of merit for gyroscope and accelerometer drift. Lower numbers mean a lower error in orientation estimation when integrating the gyroscope or accelerometer





PERFORMANCE GRADE

**GYRO BIAS INSTABILITY** 

Consumer/Hobby	> 30 °/h		
Industrial & Tactical	1 – 30 °/h		
High-end Tactical	0.1 – 1 °/h		
Navigation	0.01 - 0.1 °/h		
Strategic	0.0001 - 0.01 °/h		

output over time. It is often used to divide gyroscope performance into grades, for example; consumer, industrial, tactical, navigation or strategic. For gyroscopes, It is represented in the units of  $^{\circ}/hr$ , for accelerometers it is represented in the units of mg or ug.

**Angular Random Walk (ARW)** is the angular error buildup with time due to white noise in the angular rate. It is represented in the units of  $^{\prime}/_{hr}$ . It can also be represented as Noise Density, in the units of  $^{\prime}/_{hr}$ , by multiplying ARW by 60. More precisely, the conversation is as follows:

 $ARW(^{\circ}/h) = \frac{1}{60} \cdot FFT(^{\circ}/h/\sqrt{Hz})$ 

**Velocity Random Walk (VRW)** is the velocity error buildup with time due to white noise in acceleration expressed in  $m/s/\sqrt{hr}$ , or  $mm/s/\sqrt{hr}$ . It can also be represented as Noise Density, in the units of  $ug/\sqrt{Hz}$ 

#### Orientus, Spatial and Certus published specifications are:

- A gyroscope bias instability of 3 °/hr
- A angular random walk (ARW) of 0.24 °/√hr
- An accelerometer bias instability of 20 ug
- An accelerometer noise density of 100  $ug/\sqrt{Hz}$ , or velocity random walk (VRW) of 58 mm/s/ $\sqrt{hr}$



### Method

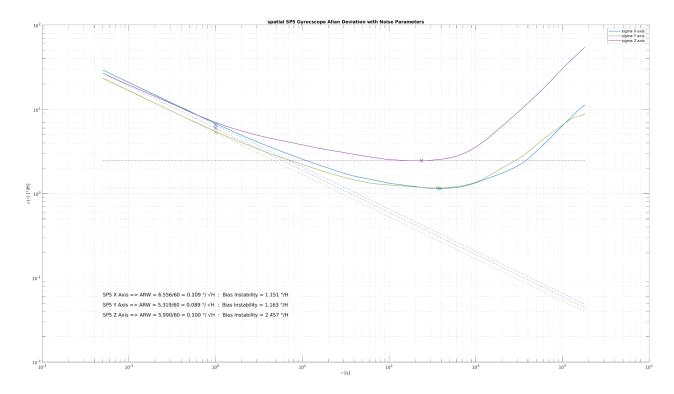
In order to carry out an Allan Variance test, a Spatial unit is situated in a low noise/vibration environment with a relatively stable temperature. The unit is mounted in a stationary fashion in an office environment overnight on an optical table. This environment is suitable to prove the performance of the units, however some artefacts may be seen on the Allan Variance plot. Tests are run for a duration of 5+ hours in order to collect sufficient data.

#### **Gyroscope Results**

The average gyroscope result across multiple axes is:

- Gyro Bias Instability = 1.59 °/hr
- Angular Random Walk (ARW) = 0.099 °/√hr

The image below is representative of the Gyroscope Allan Variance test result of a Spatial unit.



Gyroscope Allan Variance of unit of 0460046374E500C2036374B



#### **Representative Unit Gyroscope Test Results**

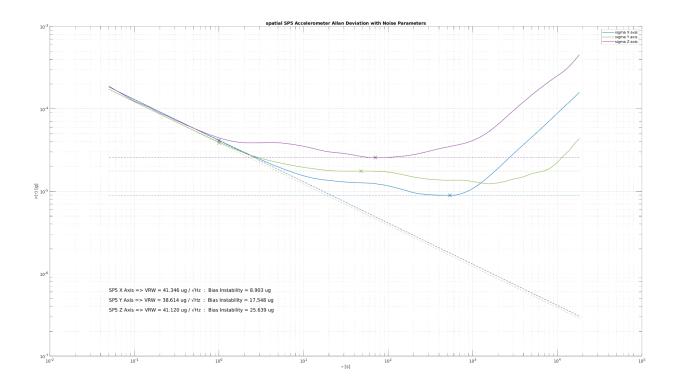
#	Unit Serial Number	Hardware version	<b>Gyro Bias</b> Instability (°/hr) (x/y/z axis)	<b>Angular Random</b> <b>Walk (°/√hr)</b> (x/y/z axis)
1	0460046374E500C2036374B	v8.0	1.15 1.16 2.46	0.109 0.089 0.100

#### **Accelerometer Results**

The average accelerometer result across multiple axes is:

- Accelerometer Bias Instability = 17.3 ug
- Velocity Random Walk (VRW) = 23.7 mm/s/√hr

The image below is representative of the Accelerometer Allan Variance test result of a Spatial unit.



Accelerometer Allan Variance of unit of 0460046374E500C2036374B



## Representative Unit Accelerometer Test Results

#	Unit Serial Number	Hardware version	Accelerometer Bias Instability (ug) (x/y/z axis)	Accelerometer Noise Density (ug/√Hz) (x/y/z axis)	Velocity Random Walk (mm/s/√hr) (x/y/z axis)
1	0460046374E500C 2036374B	v8.0	8.9 17.5 25.6	41.3 38.6 41.1	24.3 22.7 24.2



#### © ADVANCED NAVIGATION

Information in this document is provided solely in connection with Advanced Navigation products. Advanced Navigation reserves the right to make changes, corrections, modifications or improvements, to this document, and the products and services described herein at any time, without notice.

All Advanced Navigation products are sold pursuant to Advanced Navigation's terms and conditions of sale.

Purchasers are solely responsible for the choice, selection and use of the Advanced Navigation products and services described herein, and to the extent permitted by law, Advanced Navigation assumes no liability whatsoever relating to, or in anyway connected to, the choice, selection or use of the Advanced Navigation products and services described herein.

No license, express or implied, by estoppel or otherwise, to any intellectual property rights is granted under this document. If any part of this document refers to any third party products or services it shall not be deemed a license grant by Advanced Navigation for the use of such third party products or services, or any intellectual property contained therein or considered as a warranty covering the use in any manner whatsoever of such third party products or services or any intellectual property.

UNLESS OTHERWISE SET FORTH IN ADVANCED NAVIGATION'S TERMS AND CONDITIONS OF SALE ADVANCED NAVIGATION DISCLAIMS ANY EXPRESS OR IMPLIED WARRANTY WITH RESPECT TO THE USE AND/OR SALE OF ADVANCED NAVIGATION PRODUCTS INCLUDING WITHOUT LIMITATION IMPLIED WARRANTIES OF MERCHANTABILITY, FITNESS FOR A PARTICULAR PURPOSE (AND THEIR EQUIVALENTS UNDER THE LAWS OF ANY JURISDICTION), FAULT TOLERANCE OR INFRINGEMENT OF ANY PATENT, COPYRIGHT OR OTHER INTELLECTUAL PROPERTY RIGHT.

UNLESS EXPRESSLY APPROVED IN WRITING BY TWO AUTHORIZED ADVANCED NAVIGATION REPRESENTATIVES, ADVANCED NAVIGATION PRODUCTS ARE NOT RECOMMENDED, AUTHORIZED OR WARRANTED FOR USE IN MILITARY, AIR CRAFT, SPACE, LIFE SAVING, OR LIFE SUSTAINING APPLICATIONS OR HAZARDOUS OR HIGH RISK ENVIRONMENTS REQUIRING FAIL-SAFE OR FAULT TOLERANT PERFORMANCE, NOR IN PRODUCTS OR SYSTEMS WHERE FAILURE OR MALFUNCTION MAY RESULT IN PERSONAL INJURY, DEATH, OR SEVERE PROPERTY OR ENVIRONMENTAL DAMAGE. SUCH USE IS AT PURCHASERS OWN RISK, EVEN IF ADVANCED NAVIGATION KNOWS OF SUCH USE.

ADVANCED NAVIGATION PRODUCTS WHICH ARE NOT SPECIFIED AS "AUTOMOTIVE GRADE" MAY ONLY BE USED IN AUTOMOTIVE APPLICATIONS AT USER'S OWN RISK.

Resale of Advanced Navigation products with provisions different from the statements and/or technical features set forth in this document shall immediately void any warranty granted by Advanced Navigation for the Advanced Navigation product or service described herein and shall not create or extend in any manner whatsoever, any liability of Advanced Navigation.

Information in this document supersedes and replaces all information previously supplied. References to Advanced Navigation in this disclaimer includes its related bodies corporate.

© 2024 Advanced Navigation - All rights reserved.